SUPPLEMENTAL TYPE CERTIFICATE

10090936

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AVIO DESIGN GROUP INC.
d/b/a AVIO DG

305 WOODSIDE PLACE
OKOTOKS AB T1S 1L9
CANADA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.R.008
EASA.R.146

Type Certificate Holder: AIRBUS HELICOPTERS

Type: AS 350 / EC 130
AS 355

Model: AS 350 B
AS 350 B1
AS 350 B2
AS 350 B3
AS 350 BA
AS 350 D
AS 355 E
AS 355 F
AS 355 F1
AS 355 F2
AS 355 N

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 22 September 2026

Fabrice LEGAY
Section Manager
Medium & Light Rotorcraft

Task Number: 60100544
AVIO DESIGN GROUP INC. - 309335

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AS 355 NP

EC 130 B4

EC 130 T2

Description of Design Change:

Replacement Fuel Cap.

EASA Certification Basis:

The Type and OSD Certification Bases for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any impact on the noise database.

Associated Technical Documentation:

Installation of the Replacement Fuel Cap must be in accordance with Transport Canada Civil Aviation (TCCA) approved Avio Design Group Inc. (Avio dg) Modification Instruction 2052-MI-001 Revision "C", dated November 9, 2022.

Avio dg Master Data List 2052-MD-001 Revision "C", dated November 9, 2022.

Or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU/ Canada Bilateral Agreement

Limitations/Conditions:

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -

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